



Returning to the Moon: NASA's Artemis Missions

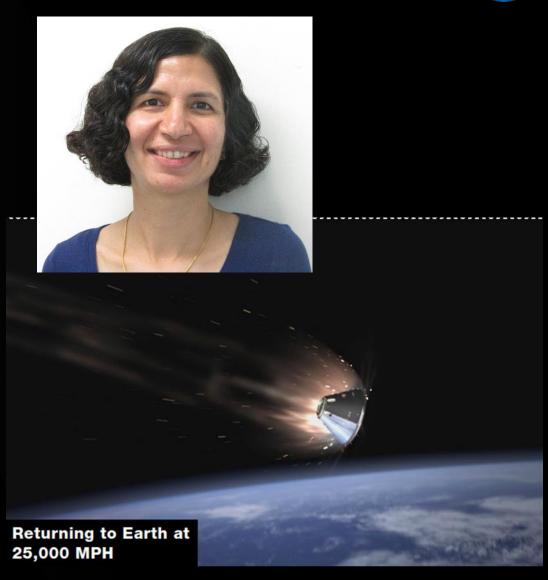
Parul Agrawal
Engineering Project Manager
NASA Ames Research Center



### PERSONAL INFORMATION



- Project Manager for Orion (Multi-Purpose Crew Vehicle) operations at NASA Ames Research Center since 2018
  - Orion is one of the elements of NASA's Artemis program with objective to land the first woman and next man on Moon by 2024
- Project Manager for Science mission proposals
- Prior to current role, I spent 11 years in Entry Systems division designing, developing and testing heatshield materials and systems. I had the opportunity to work on many different projects at NASA.
  - Mission concept studies to send probes to Ice giants
  - Thermal analysis of Entry vehicles
  - Asteroid entry on Earth and their break-up





## ARTEMIS

Twin sister of Apollo and goddess of the Moon in Greek mythology, Artemis is the torch-bringer personifying our path to the Moon. During the next era of human exploration, we will discover life-saving, Earth-changing science and technology along the way.

NASA's goal is to land the first woman and first person of color on the Moon and return them safely to Earth. When the Artemis astronauts land on the lunar surface, they will step into the future, bringing all of humanity with them.

## Why Artemis?

## DISCOVER

With Artemis, we're building on more than 50 years of exploration experience to reignite America's passion for discovery







Locate concentrations of water ice that could eventually be harvested to sustain human exploration on the Moon, Mars — and beyond. VIPER represents the first resource mapping mission on another celestial body.

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## **OPPORTUNITY**

Artemis missions enable a growing lunar economy by fueling new industries, supporting job growth, and furthering the demand for a skilled workforce.



## **INSPIRATION**

With Artemis missions, NASA will land an inspirational crew on the Moon, using innovative technologies to expand

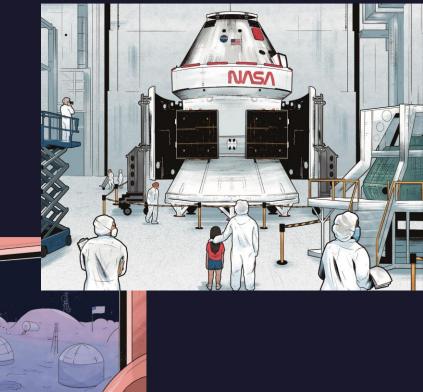
human exploration.



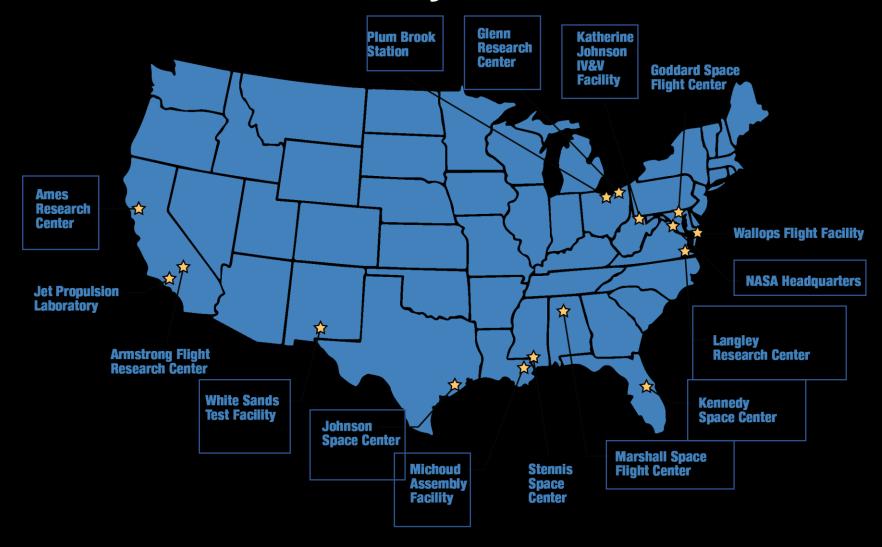


# YOU ARE GOING

A STORY ABOUT ARTEMIS



### **Key Artemis Contributions by NASA Centers**



## HOW WE GET THERE!







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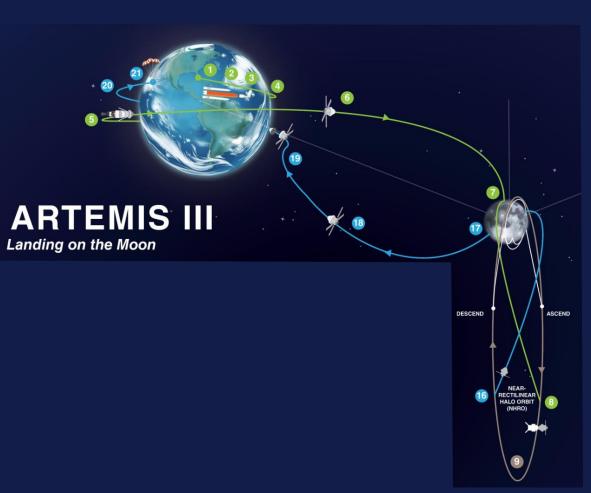


## HOW WE GET THERE!













## MOON ROCKET



www.nasa.gov/sls

#### **ORION STAGE ADAPTER**

The adapter carries small satellites to deep space where they conduct world-class science for pennies on the dollar.

### INTERIM CRYOGENIC PROPULSION STAGE (ICPS)

One RL10 engine provides 24,750 pounds of thrust to send Orion to the Moon.

### LAUNCH VEHICLE STAGE ADAPTER

The adapter connects the 27.5-foot diameter core stage to the 16.5-foot diameter ICPS and partially encloses the ICPS in-space stage.

#### CORE STAGE

The 212-foot tall core stage holds-733,000 gallons of propellant to power four RS-25 engines for eight minutes, sending the rocket soaring to space at 17,000 miles per hour.

#### SOLID ROCKET BOOSTERS

Each 17-story-tall booster generates 3.6 million pounds of thrust, providing 75 percent of total thrust during the SLS rocket's first two minutes of flight.

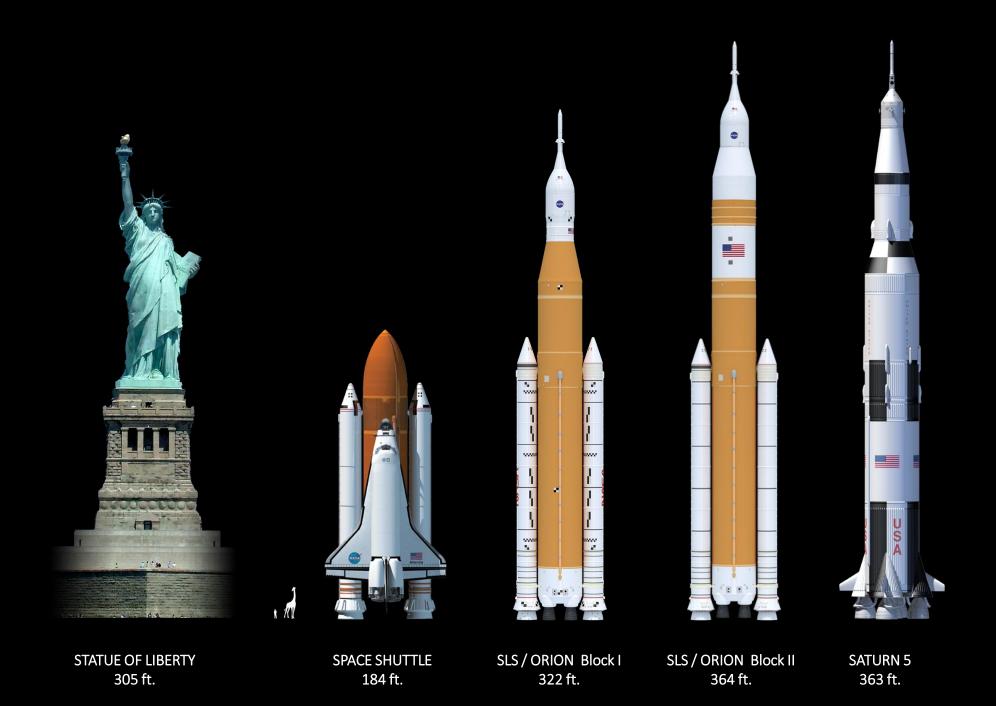
#### **FOUR RS-25 ENGINES**

As the most efficient engines ever built, the engines provide a total of two million pounds of thrust for launch and ascent to space.

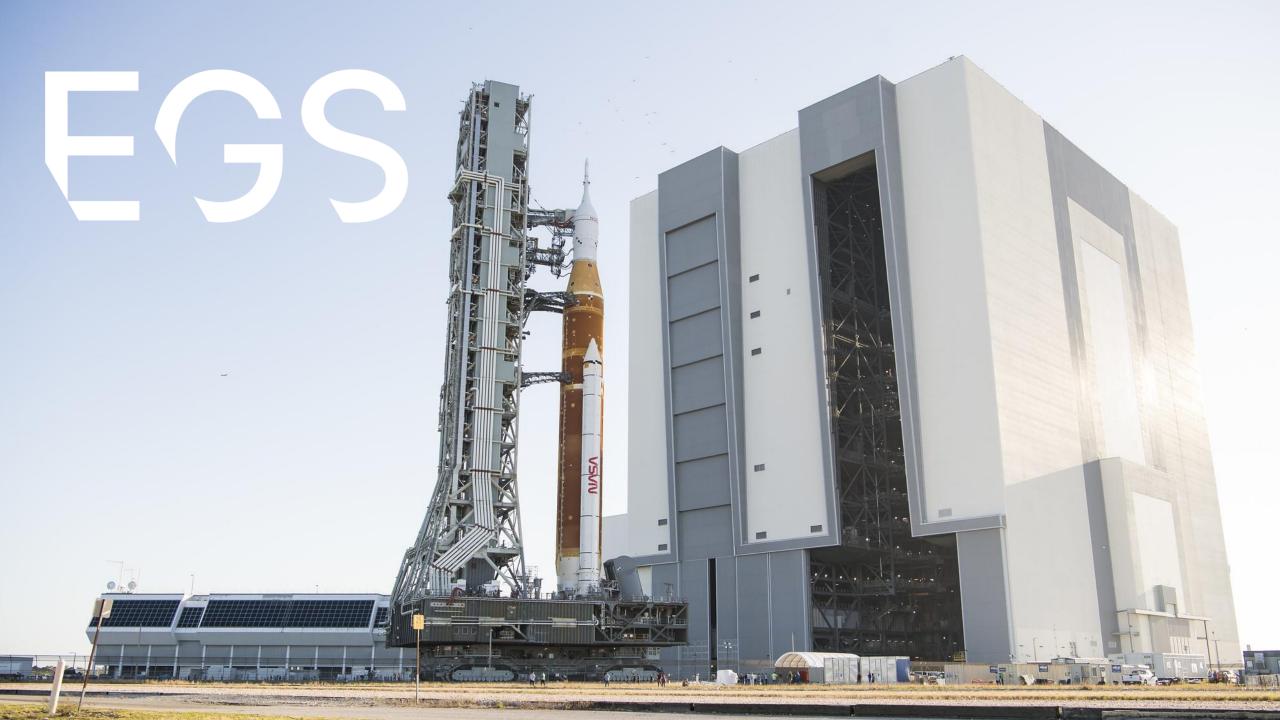


Once in space, need **24,750 pounds** of thrust to send Orion to the Moon!

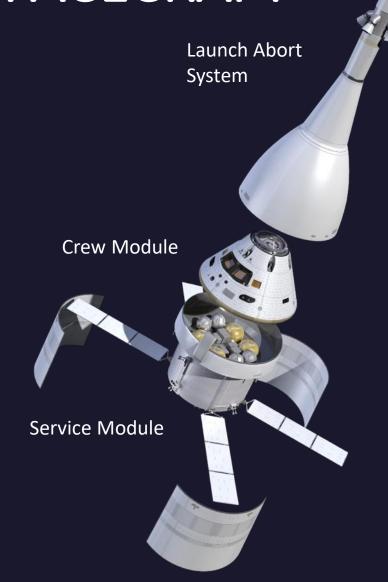
Total of **5.6 Million pounds** of thrust to lift 59,525 pounds from the Earth's surface, escape gravity.







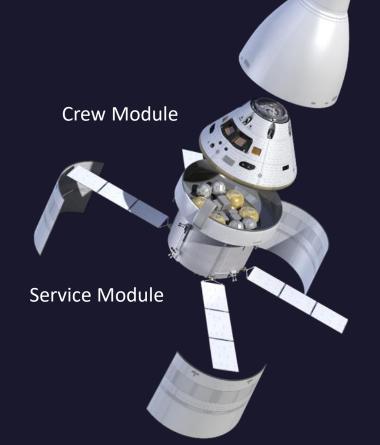
# THE ORION SPACECRAFT

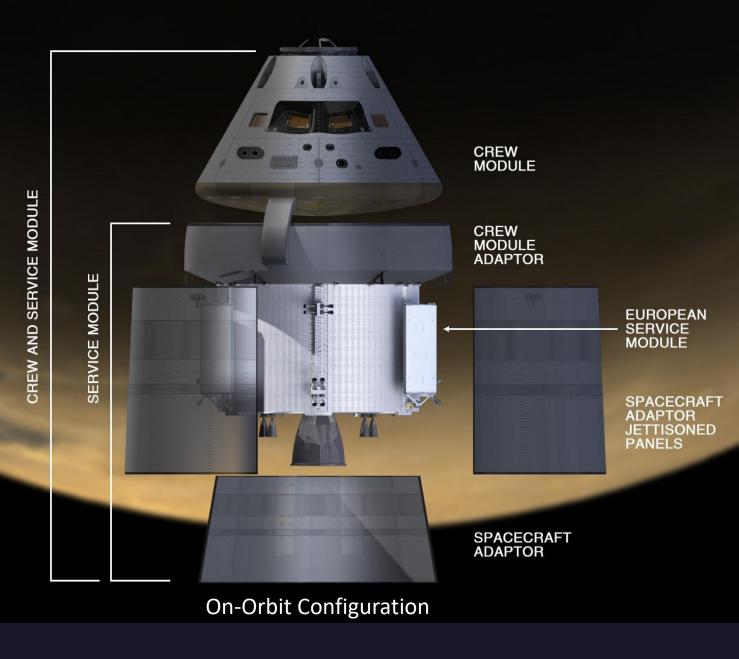




# THE ORION SPACECRAFT

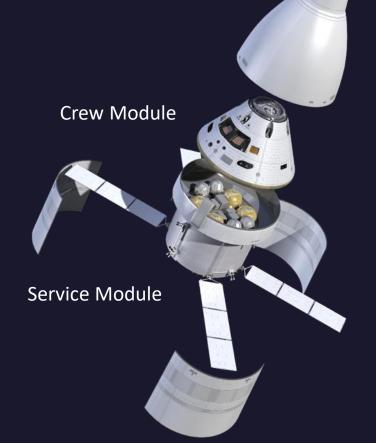
Launch Abort System

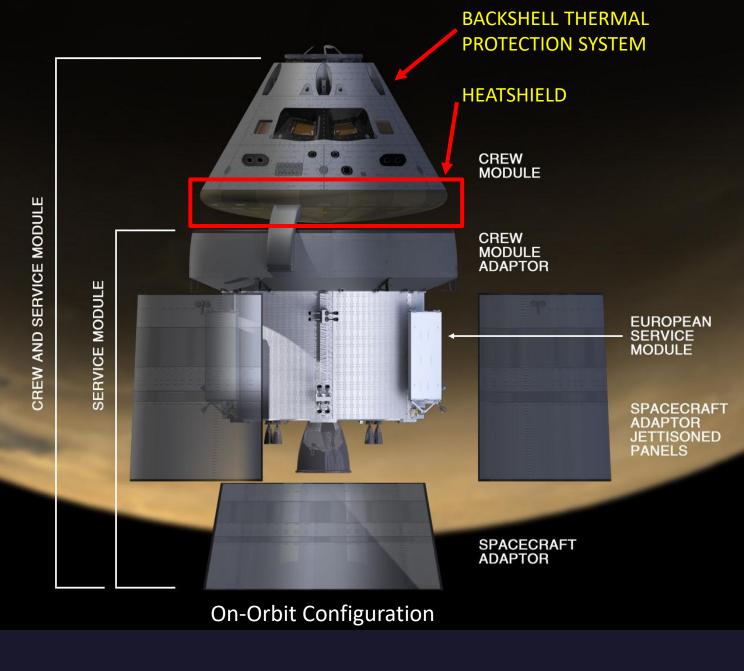




# THE ORION SPACECRAFT

Launch Abort System





## **Artemis I Payloads**

Science and technology investigations and demonstrations paving the way for future, deep space human exploration



### **Moonikin Campos**

The Moonikin is a male-bodied manikin previously used in Orion vibration tests. Campos will occupy the commander's seat inside and wear an Orion Crew Survival System suit



### **Radiation Sensors**

There will be three types of sensors, including the ESA Active Dosimeters, Hybrid Electronic Radiation Assessor, and the Radiation Area Monitor.



### **MARE**

Radiation shielding Personal Protection Equipment (radiation vest) for astronauts.



### Technology Payload (CITP) Creates an interactive

experience between Orion and the public during the mission



**Bio-Experiment-1** 

**Crew Interface** 

Battery-powered life sciences payload for biology research beyond low-Earth orbit (LEO)



ArgoMoon



LunaH-Map



**EQUULEUS** 



**OMOTENASHI** 



LunIR



Near-Earth Asteroid Scout (NEA Scout)



**Lunar IceCube** 



**BioSentinel** 



**Team Miles** 



**CuSP** 











The capsule must survive external **temperatures as high as 5,000** °F! That's half the surface temperature of the Sun!!!!

Our heat shield technology helps keep the inside of the capsule (where the humans will live!) in the mid-70 °F

This is the largest ablative heatshield in the world at **16.5 ft** in diameter!











Orion is equipped with 11 parachutes that are deployed to slow the vehicle from 325 mph to 20 mph for a soft landing in the ocean.







## ARTEMIS-1 Successes!

SUCCESSFUL LAUNCH OF MOST POWERFUL ROCKET IN THE WORLD







## ORION IN DISTANT RETROGRADE ORBIT CAPTURES MOON TRANSIT IN FRONT OF EARTH



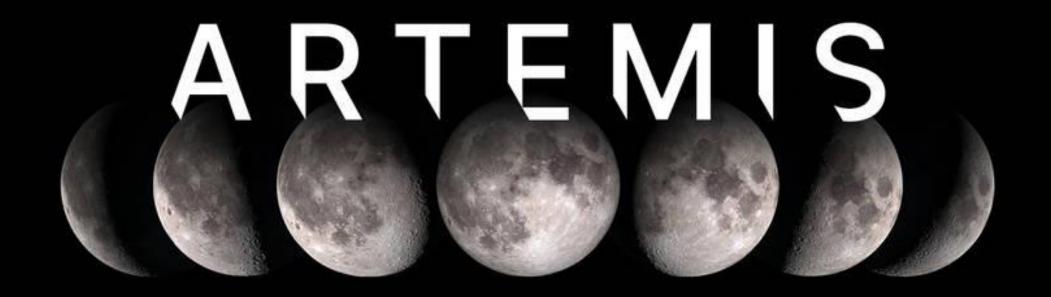
## ARTEMIS-1 Successes!

SUCCESSFUL RECOVERY IN THE PACIFIC OCEAN









QUESTIONS?

## VALUABLE LUNAR SCIENCE



Study of Planetary Processes



Understanding Volatile Cycles



Impact History of Earth-Moon System



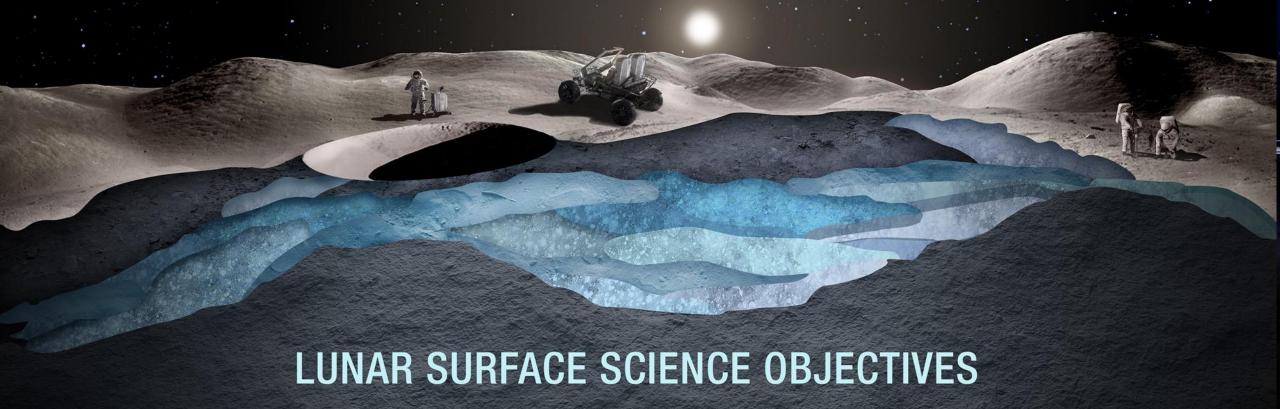
Record of the Ancient Sun



Fundamental Lunar Science



Platform to Study the Universe



## Mission Needs Drive Design

**LOW EARTH RETURN** 

3 HOURS 3,000°F 17,500 MPH 250 MILES **LUNAR RETURN** 

3 DAYS 5,200°F 24,700 MPH 240,000 MILES MARS RETURN

9 MONTHS 6,200°F 26,800 MPH 39,000,000 MILES



## Artemis: a Foundation for Deep Space Exploration









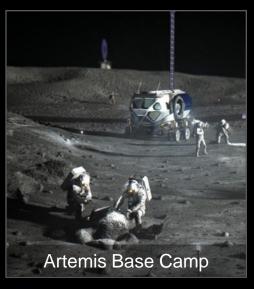




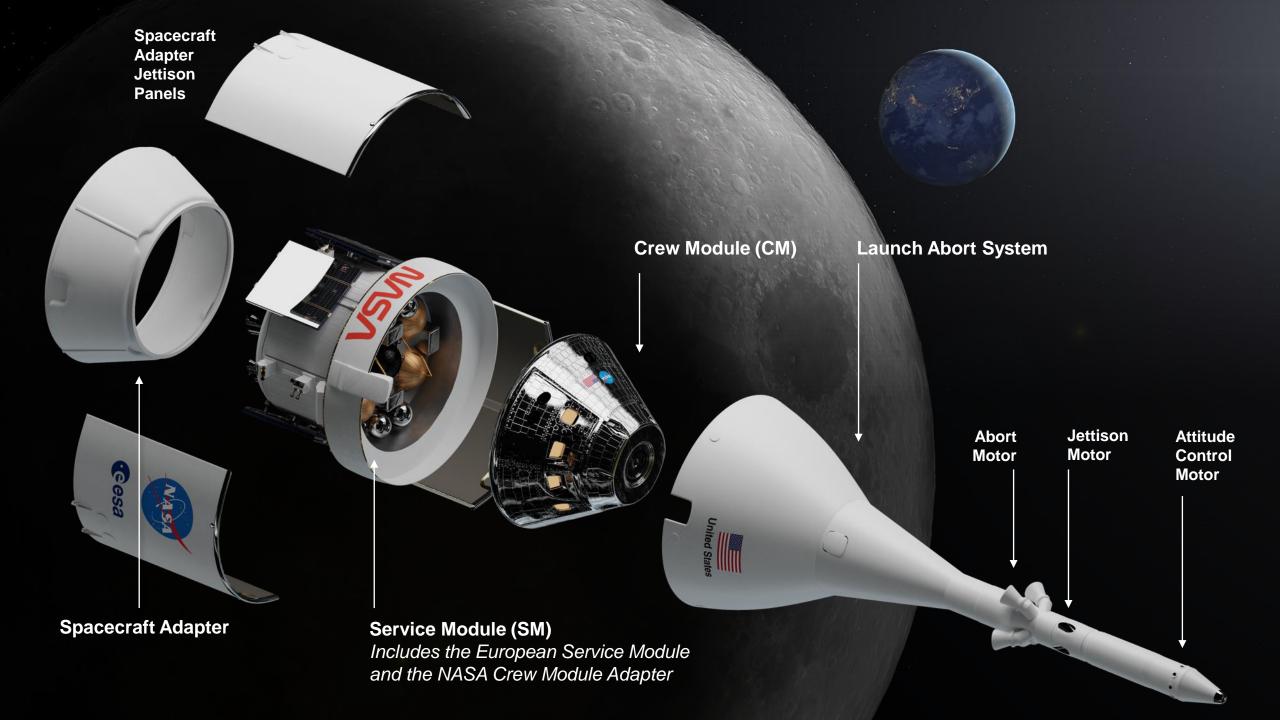












## Orion Quick Facts



Performance	Height
Number of crew 4	Crew module + 26 ft.
Mission Duration up to 21 days	service module
	Orion stack (launch abort 67 ft.
	system + crew module
	+ service module)
	SLS Block 1 Configuration 322 ft.
	(Orion + SLS stack)
Trans-Lunar Insertion Mass	Post-Trans Lunar Insertion Mass
Artemis I 53,000 lbs.	Artemis I 51,500 lbs.
Artemis II 58,000 lbs.	Artemis II 57,000 lbs.
	Usable Propellant 19,000 lbs.
	Total Change in Velocity (ΔV) with Fully Loaded
Gross Liftoff Weight	Propellant Tank
Artemis I 72,000 lbs.	Artemis I 53,000 lbs.
Artemis II 78,000 lbs.	Artemis II 58,000 lbs.